



**THE FLAME PROPAGATION OF VARIOUS FUELS IN A PARTICULAR COMBUSTION CHAMBER OF 4.-VALVE ENGINES**



Zoran Jovanovic<sup>1†</sup> --- Zoran Masonic<sup>2</sup>

<sup>1,2</sup>University of Belgrade, Institute of Nuclear Sciences, "Vinca", Laboratory for IC Engines and Motor Vehicles, Belgrade, Serbia

**ABSTRACT**

*In this paper some results concerning flame propagation of various fuels in a particular combustion chamber with four tilted valves were elucidated. Flame propagation was represented by the evolution of spatial distribution of temperature in various cut-planes within combustion chamber while the flame front location was determined by dint of zones with maximum temperature gradient. Results presented are only a small part of broader on-going scrutinizing activity in the field of multidimensional modeling of reactive flows in combustion chambers with complicated geometries encompassing various models of turbulence, different fuels and combustion models. In the case of turbulence two different models were applied i.e. standard k-ε model of turbulence and k-ξ-f model of turbulence. In this paper flame propagation results were analyzed and presented for two different hydrocarbon fuels, such as CH<sub>4</sub> and C<sub>8</sub>H<sub>18</sub>, for both turbulence models applied. In the case of combustion all differences ensuing from different turbulence models, obvious for non-reactive flows are annihilated entirely. Namely the interplay between fluid flow pattern and flame propagation is invariant as regards turbulence models and fuels applied. Namely the interplay between fluid flow pattern and flame propagation is entirely invariant as regards fuel variation indicating that the flame propagation through unburned mixture of CH<sub>4</sub> and C<sub>8</sub>H<sub>18</sub> fuels is not chemically controlled.*

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**Keywords:** Automotive flows, Flame propagation, Combustion modeling, Turbulence modeling, CFD, Fuel variation, 4.-valve engine

### Contribution/ Originality

The paper's primary contribution is finding that combustion chemistry of fuel considered (CH<sub>4</sub>/C<sub>8</sub>H<sub>18</sub>) is of no importance for flame propagation in a particular combustion chamber. In spite of the fact that in this case flame propagation is fully controlled by turbulent diffusion no effect due to turbulence model change is observed.

## 1. INTRODUCTION

It is known for a long time that various types of organized flows in combustion chamber of IC Engines are of predominant importance for combustion particularly as regards flame front shape and its propagation. Some results related to the isolated or synergic effect of squish and swirl on flame propagation in various combustion chamber layouts are already analysed and published [1] but results concerning the isolated or combined effect of the third type of organized flow i.e. tumble are relatively less presented [2]. From the theory of turbulence is known that vortex filament subjected to compression reduces its length and promotes rotation around its axis yielding the movement on the larger scale ("spin-up" effect). It can be presumed that tumble pursues the same rule i.e. the destruction of formed and expressive tumble during compression stroke generates the higher turbulence intensity and larger integral length scale of turbulence in the vicinity of TDC contributing to the flame kernel formation period reduction and faster flame propagation thereafter. The aforementioned logic imposes the conclusion that the most beneficial fluid flow pattern adjacent to BDC is well shaped high intensity tumble flow.

## 2. MODEL AND COMPUTATIONAL METHOD

The analysis of this type is inherent to multidimensional numerical modelling of reactive fluid flow and therefore it is quite logical to apply such a technique particularly due to fact that it is the only technique that encompasses the valve/port geometry layout in an explicit manner. In lieu of the fact that, in its essence, multidimensional models require initial and boundary conditions only their applications is fairly complicated and imply some assumptions and simplifications [3]. The full 3D conservation integral form of unsteady equations governing turbulent motion of non-reactive mixture of ideal gas is solved on a fine computational grid with moving boundaries in physical domain by dint of AVL FIRE code [4]. In this case the numerical solution method is based on a fully conservative finite volume approach. For the solution of a recast linear system of equations, a conjugate gradient type of solver (CGS) is used. Two different models of turbulence were used. The first one is nearly forty years old k- $\epsilon$  model based on Boussinesq's assumption which is, certainly, the most widely used model for engineering computations. The second one is k- $\xi$ -f model of turbulence i.e. eddy-viscosity model based on Durbin's elliptic relaxation concept [5 - 7]. This model solves a transport equation for the velocity scale ratio  $\xi$  instead of imaginary turbulent normal stress component. In addition, the pertinent hybrid boundary conditions were applied.

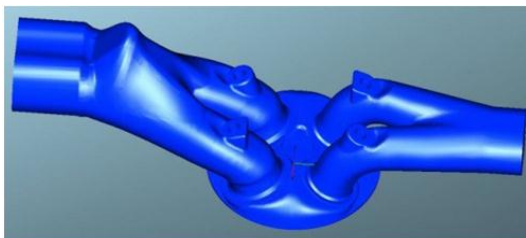
### 3. RESULTS AND DISCUSSION

#### 3.1. Fuel Variation

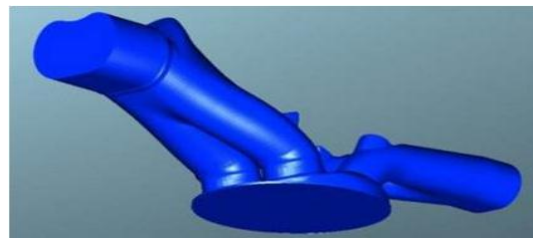
The analysis of relevant results was based on a fairly complex geometry layout presented in figures 1. and 2. Obviously, combustion chamber is constrained with dual intake and exhaust valves. The basic block data sheet consists of bore/stroke ratio = 80/81.4mm, squish gap=1.19 mm, engine speed RPM = 5500 min<sup>-1</sup> and mixture quality  $\lambda=1$ . In addition, maximum valve lift is 6.95mm for intake valves and 6.63 mm for exhaust valves while the other geometrical data (relative location, valve shape etc.) could be seen in fig.1 and 2. It should be stated that results presented in this paper were obtained by dint of AVL FIRE code [4]. These results were carefully selected from bunch of results (more than 3000 plots) obtained for various combinations of fuel, turbulence and combustion, as it is set forth in table 1, below.

**Table-1.** Combination of fuel, turbulence and combustion model

4-valve engine (fig.1 and 2.)			
Bore/Stroke	80/81.4 mm		
Rpm	5500 rpm		
Squish gap	1.19 mm		
Mixture quality ( $\lambda$ )	1		
	<i>Fuel used</i>	<i>Turbulence model</i>	<i>Combustion model</i>
	Petrol	k- $\xi$ -f	Probability Density
	Petrol	k- $\epsilon$	Probability Density Function
	N-octane	k- $\epsilon$	Magnusen-Hjaertager
	N-octane	k- $\xi$ -f	Magnusen-Hjaertager
	N-octane	k- $\epsilon$	Probability Density
	CH4	k- $\epsilon$	Magnusen-Hjaertager
	CH4	k- $\xi$ -f	Magnusen-Hjaertager
	CH4	k- $\xi$ -f	Probability Density Function
	H2	k- $\epsilon$	Magnusen- Hjaertager



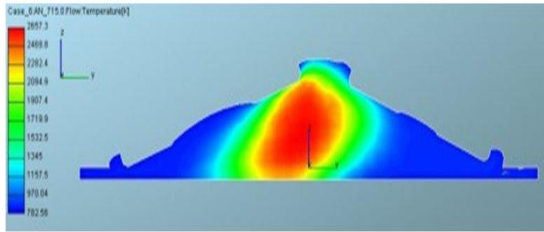
**Fig-1.** Perspective view of the combustion chamber geometry layout with 4-valves (upper view)



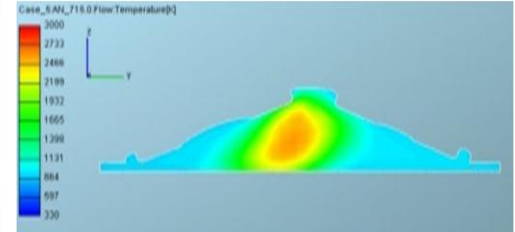
**Fig-2.** Perspective view of the chamber geometry layout with 4-valves (bottom view)

The flame propagation through unburned mixture of two different hydrocarbon fuels, such as CH<sub>4</sub> and C<sub>8</sub>H<sub>18</sub> was analyzed by dint of the evolution of spatial distribution of temperatures, represented in form of iso-contours in six cut-planes passing through various parts of combustion chamber geometry layout depicted in fig. 1 and 2. The flame propagation i.e. the evolution of spatial distribution of temperature in the first cut-plane for C<sub>8</sub>H<sub>18</sub> and CH<sub>4</sub> fuels is shown in

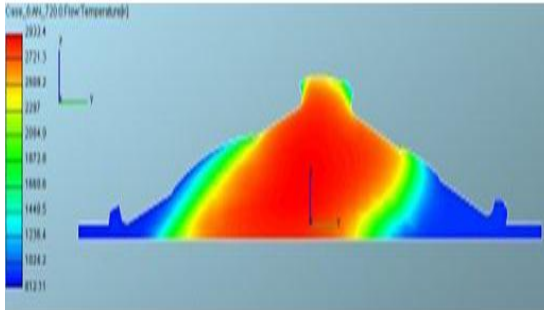
figs. 3.-10.



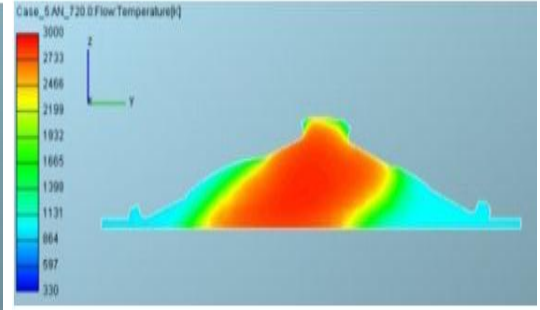
**Figure-3.** Spatial distribution of temperature in x-z plane, y=const. (1.16) at 355 deg. ATDC, k-ε, C8H18



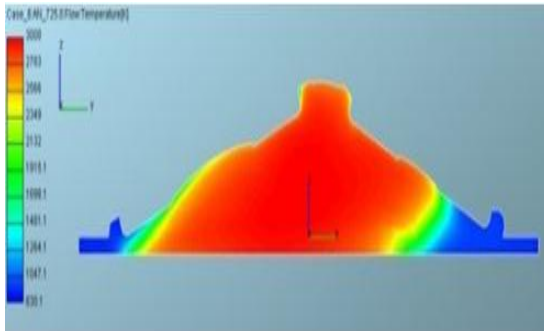
**Figure-4.** Spatial distribution of temperature in x-z plane, y=const. (1.16) at 355 deg. ATDC, k-ε, CH4



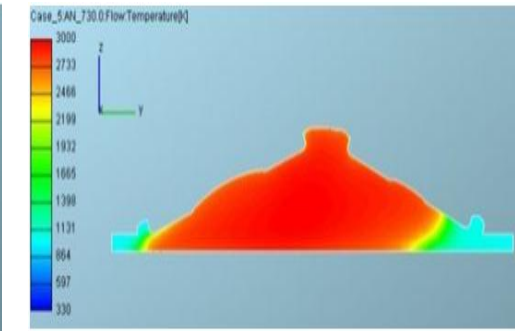
**Figure-5.** Spatial distribution of temperature in x-z plane, y=const. (1.16) at 360 deg. ATDC, k-ε, C8H18



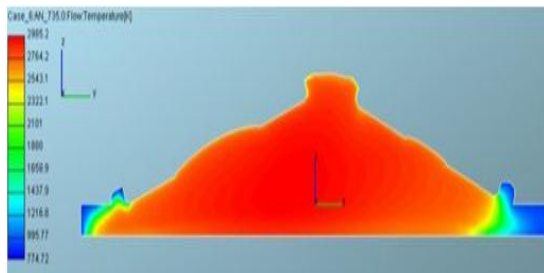
**Figure-6.** Spatial distribution of temperature in x-z plane, y=const. (1.16) at 360 deg. ATDC, k-ε, CH4



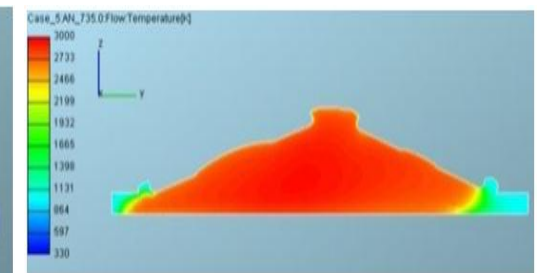
**Figure-7.** Spatial distribution of temperature in x-z plane, y=const. (1.16) at 365 deg. ATDC, k-ε, C8H18



**Figure-8.** Spatial distribution of temperature in x-z plane, y=const. (1.16) at 365 deg. ATDC, k-ε, CH4



**Figure-9.** Spatial distribution of temperature in x-z plane, y=const. (1.16) at 370 deg. ATDC, k-ε, C8H18



**Figure-10.** Spatial distribution of temperature in x-z plane, y=const. (1.16) at 370 deg. ATDC, k-ε, CH4

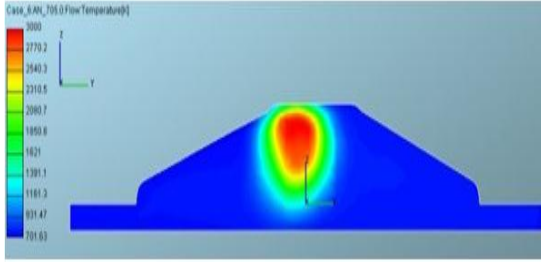


Figure-11. Spatial distribution of temperature in x-z symmetry plane, y=0.0 at 345 deg. ATDC, k-ε, C8H18

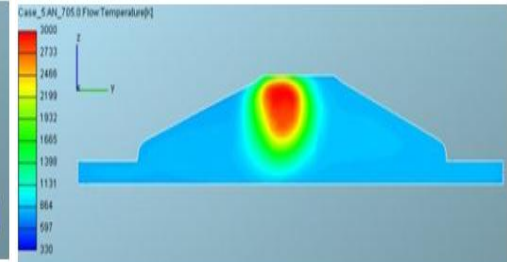


Figure-12. Spatial distribution of temperature in x-z symmetry plane, y=0.0 at 345 deg. ATDC, k-ε, CH4

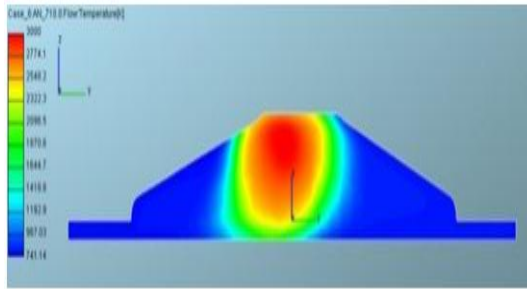


Figure-13. Spatial distribution of temperature in x-z symmetry plane, y=0.0 at 350 deg. ATDC, k-ε, C8H18

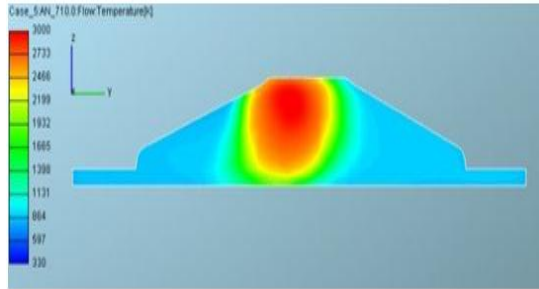


Figure-14. Spatial distribution of temperature in x-z symmetry plane, y=0.0 at 350 deg. ATDC, k-ε, CH4

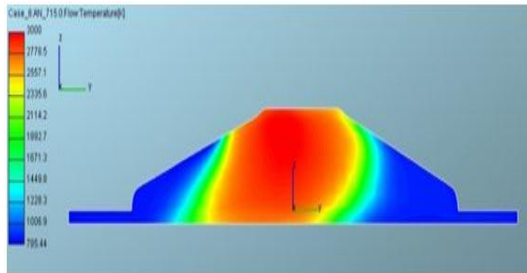


Figure-15. Spatial distribution of temperature in x-z symmetry plane, y=0.0 at 355 deg. ATDC, k-ε, C8H18

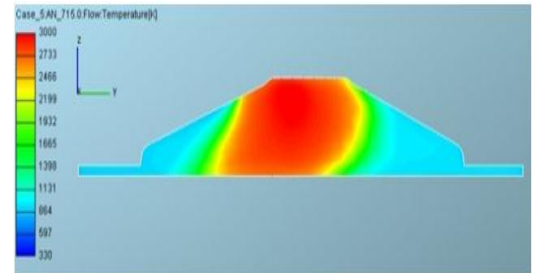


Figure-16. Spatial distribution of temperature in x-z symmetry plane, y=0.0 at 355 deg. ATDC, k-ε, CH4

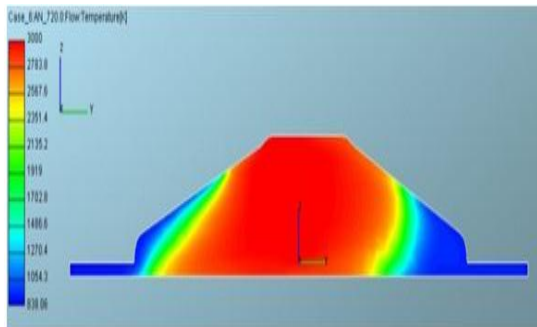


Figure-17. Spatial distribution of temperature in x-z symmetry plane, y=0.0 at 360 deg. ATDC, k-ε, C8H18

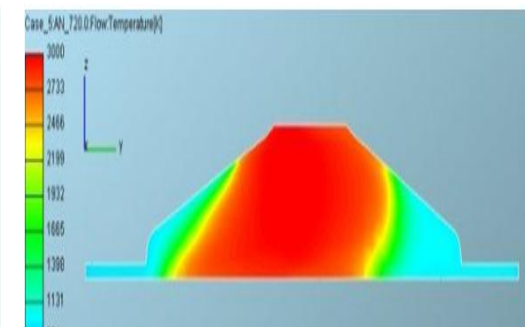
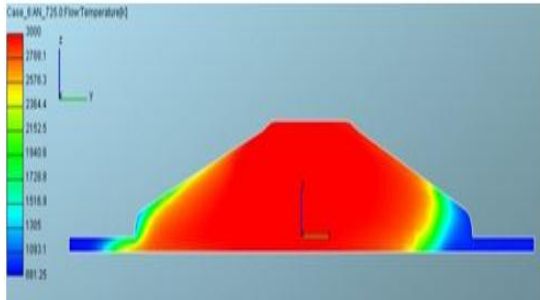
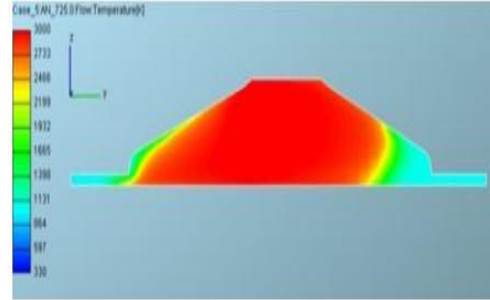


Figure-18. Spatial distribution of temperature in x-z symmetry plane, y=0.0 at 360 deg. ATDC, k-ε, CH4



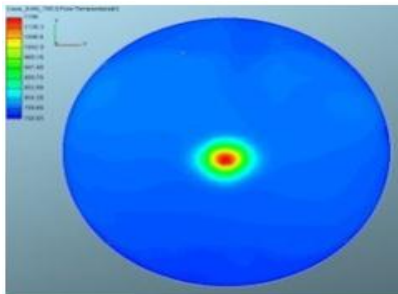


**Figure-19.** Spatial distribution of temperature in x-z symmetry plane, y=0.0 at 365 deg. ATDC, k-ε, C8H18

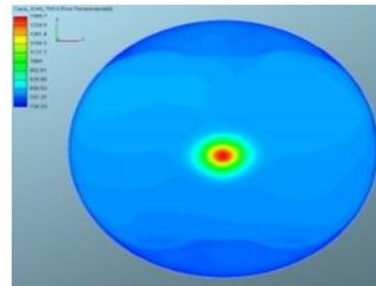


**Figure-20.** Spatial distribution of temperature in x-z symmetry plane, y=0.0 at 365 deg. ATDC, k-ε, CH4

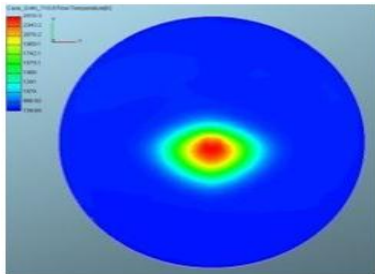
The flame propagation i.e. the evolution of spatial distribution of temperature in the sixth cut-plane for C8H18 and CH4 fuels is shown in figs. 21.-30.



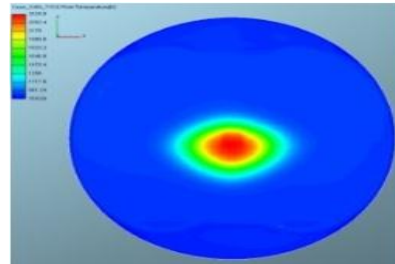
**Figure-21.** Spatial distribution of temperature in x-y plane passing through squish zone at 345 deg. ATDC, k-ε, C8H18



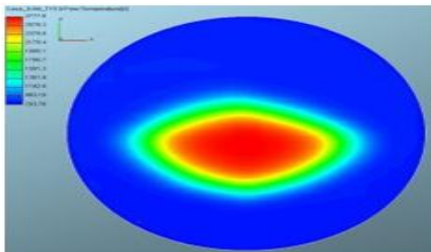
**Figure-22.** Spatial distribution of temperature in x-y plane passing through squish zone at 345 deg. ATDC, k-ε, CH4



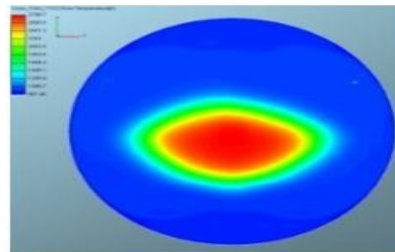
**Figure-23.** Spatial distribution of temperature in x-y plane passing through squish zone at 350 deg. ATDC, k-ε, C8H18



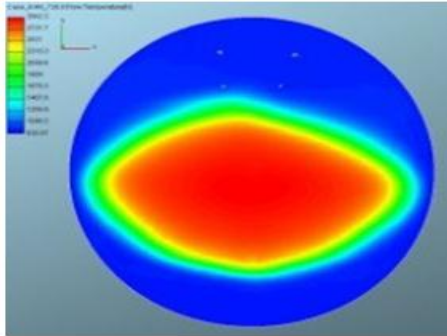
**Figure -24.** Spatial distribution of temperature in x-y plane passing through squish zone at 350 deg. ATDC, k-ε, CH4



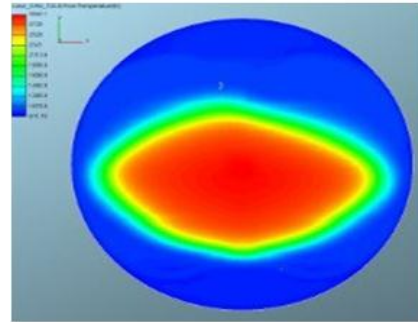
**Figure-25.** Spatial distribution of temperature in x-y plane passing through squish zone at 355 deg. ATDC, k-ε, C8H18



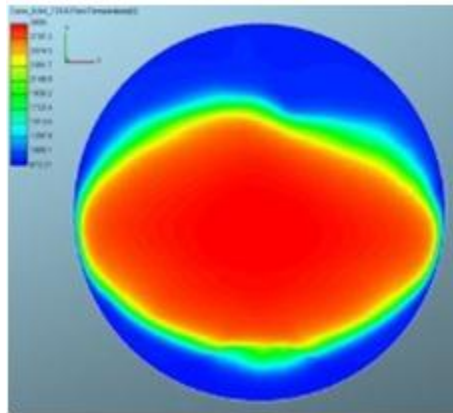
**Figure-26.** Spatial distribution of temperature in x-y plane passing through squish zone at 355 deg. ATDC, k-ε, CH4



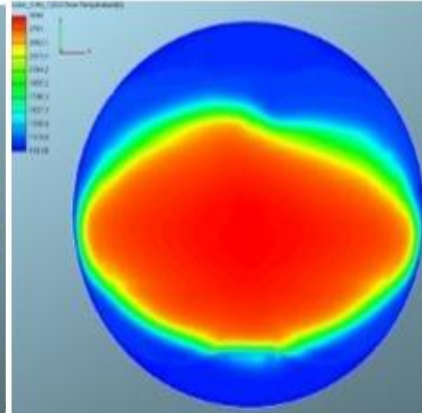
**Figure-27.** Spatial distribution of temperature in x-y plane passing through squish zone at 360 deg. ATDC, k-ε, C8H18



**Figure-28.** Spatial distribution of temperature in x-y plane passing through squish zone at 360 deg. ATDC, k-ε, CH4



**Figure-29.** Spatial distribution of temperature in x-y plane passing through squish zone at 365 deg. ATDC, k-ε, C8H18



**Figure-30.** Spatial distribution of temperature in x-y plane passing through squish zone at 365 deg. ATDC, k-ε, CH4

Obviously neither differences in flame propagation nor in flame front shape in all cut-planes were encountered for both hydrocarbon fuels used (C8H18 and CH4) yielding the conclusion that flame front shape and its displacement are not chemically controlled but controlled by dint of turbulent diffusion i.e. by high intensity of turbulence and cascade process of tearing or breaking up large vortices into smaller ones and their dissipation into heat.

#### 4. CONCLUSIONS

The fluid flow pattern during induction and compression in a particular combustion chamber geometry of 4-valve engine is extremely complex and entirely three-dimensional. The modeling of turbulence strongly affects the evolution of fluid flow pattern and spatial distribution of kinetic energy of turbulence in 4-valve engines but only in the case of non-reactive flow. In general, k-ε model of turbulence generates higher values of kinetic energy of turbulence over the broader part of the chamber than corresponding k-ξ-f model of turbulence. *In the case of combustion all differences ensuing from turbulence model variation encountered in the case of non-reactive fluid flow are annihilated entirely.* Namely the interplay between fluid flow pattern encountered (“flame dominated fluid flow”) and flame propagation is invariant as

regards both turbulence models applied. On the contrary, such a conclusion is not valid either in the case of “squish dominated flows” or in the case of “coincident flow” [1]. The flame front shape and its displacement in IC Engine combustion chamber with strong macro flows are entirely invariant as regards fuel variation tested for both turbulence models indicating that flame propagation is not chemically controlled but controlled by dint of turbulent diffusion. Heat release due to chemical reactions on right hand side of energy equation is of no importance for flame front shape and its displacement presuming that this invariance is valid for broad range of hydrocarbon fuels.

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